

*Seventh Meeting- UNILATERAL PROBLEMS IN
STRUCTURAL ANALYSIS Palmanova, June 16-19 2010*

Dynamic delamination phenomena in composite structures

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DYNAMIC FRACTURE MECHANICS

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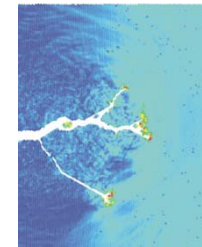
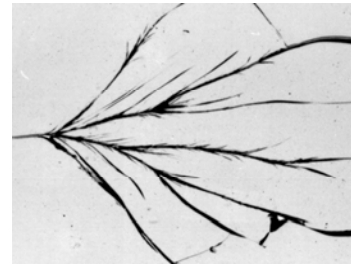
FE MODEL

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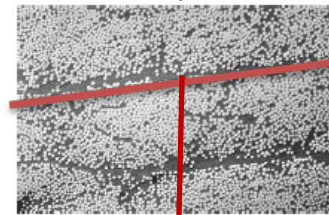
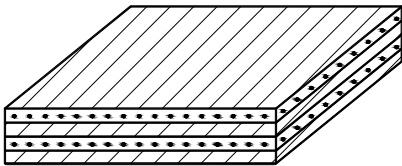
CONCLUSIONS

MONOLITIC MATERIALS

- ❑ Crack Branching phenomena
- ❑ Crack speeds are limited
- ❑ Unknown path of the crack



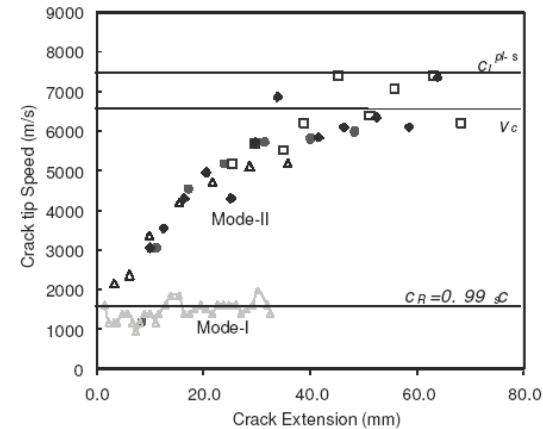
COMPOSITE STRUCTURES



Weak plane

- ❑ High crack speed
- ❑ Crack constrained along the interfaces

Ravi-Chandar and Knauss, *Int J Fract*, 1984



(Rosakis, A.J., "Intersonic shear cracks and fault ruptures propagation", *Advances in Physics*, 2002)



DYNAMIC CRACK GROWTH MODELING

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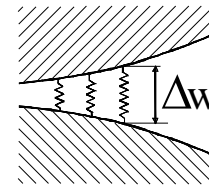
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❏ Cohesive modeling

➔ Interface elements are introduced at the crack region

➔ Damaged constitutive relationship is required

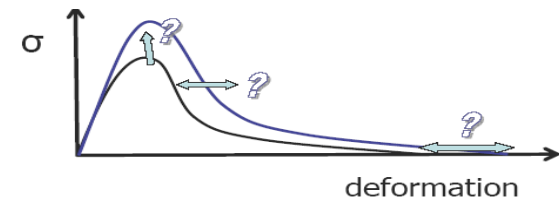


❏ Fracture Mechanics approaches

➔ Static analyses:
(the time dependence is neglected “a priori”)

➔ Steady state crack growth approaches:
(A moving reference system is fixed at the tip, crack tip speed is constant)

➔ Unsteady models :
Full Time dependence , inertial forces,
loading rate,....



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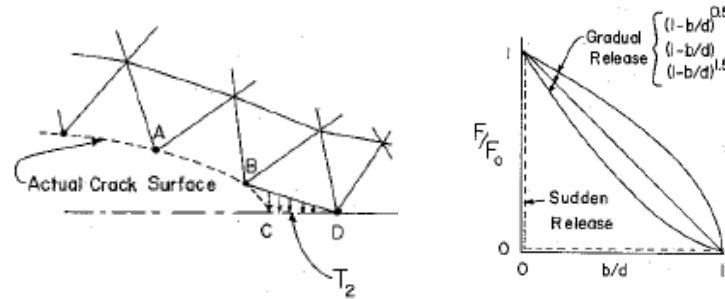
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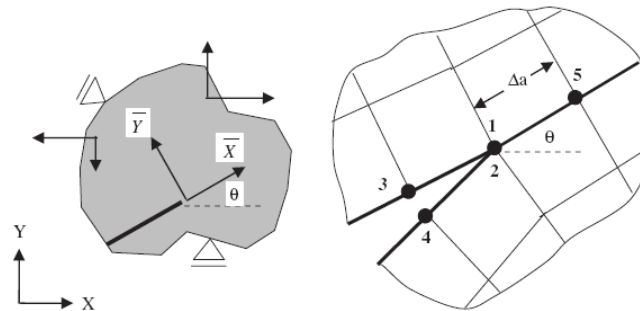
Node release technique

➔ Gradual release of the nodal forces behind the crack tip



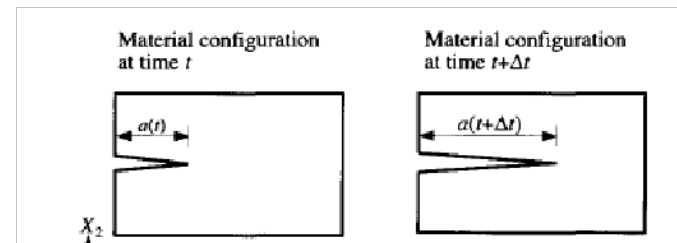
Virtual crack closure methods

➔ The ERR is evaluated by the mutual work at the crack tip and behind the crack tip



Moving mesh methodology

➔ The nodes are moved to predict changes of the geometry produced by the crack motion



MOTIVATION OF THE WORK AND SUMMARY

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AIM OF THE WORK

 Propose a generalized modeling based on Fracture mechanics and moving mesh methodology to predict the dynamic behavior of composite laminated structures

SUMMARY

 Review the main equations of the ALE formulation in view of the Dynamic Fracture Mechanics approach

 Evaluate the specialized expressions of the ERR by the use of the decomposition methodology of the J-integral and propose a proper mixed mode crack toughness criterion

 Develop the finite element implementation. Propose validation by means of comparisons with experimental data and a parametric study to analyze dynamic crack behavior (i.e. crack arrest phenomena, allowable tip speeds, cracks interaction)



BASICS OF MOVING MESH STRATEGY: ARBITRARY-LAGRANGIAN EULERIAN FORMULATION

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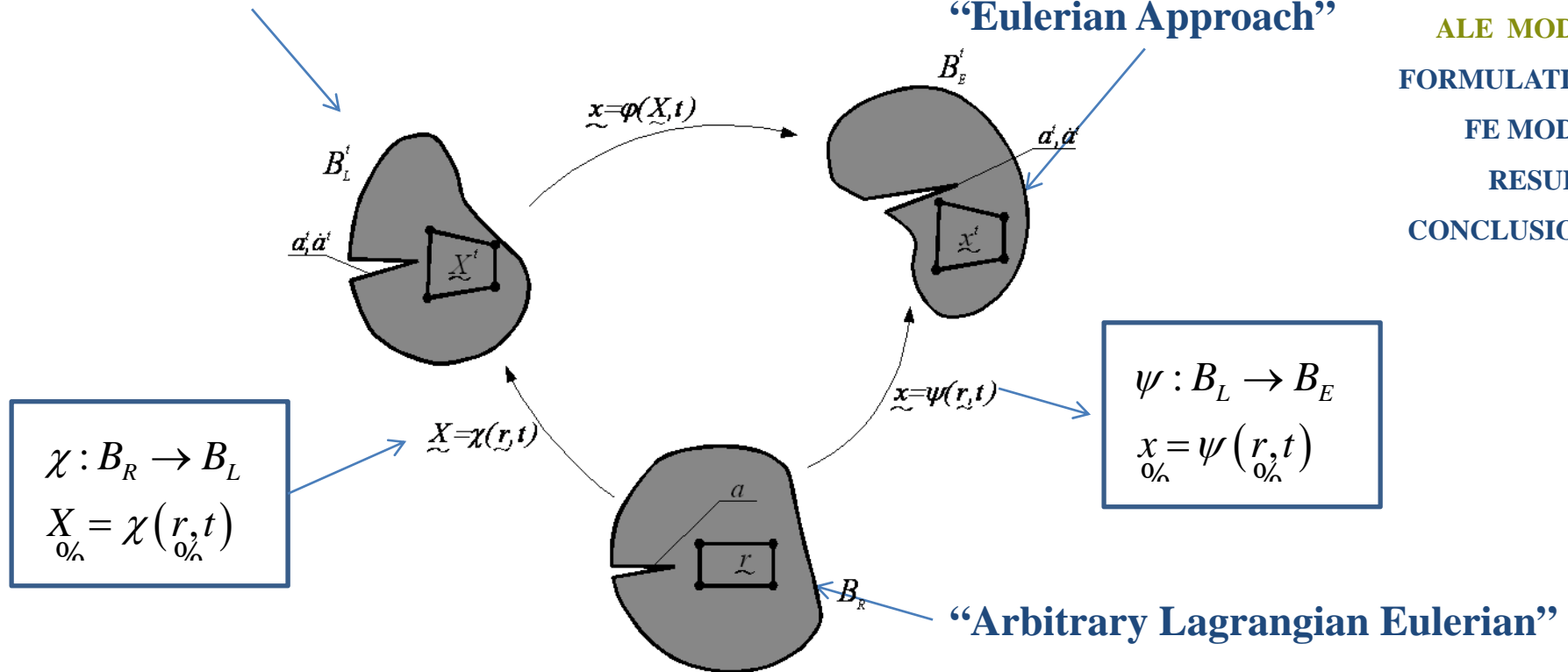
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“Lagrangian Approach”

“Eulerian Approach”



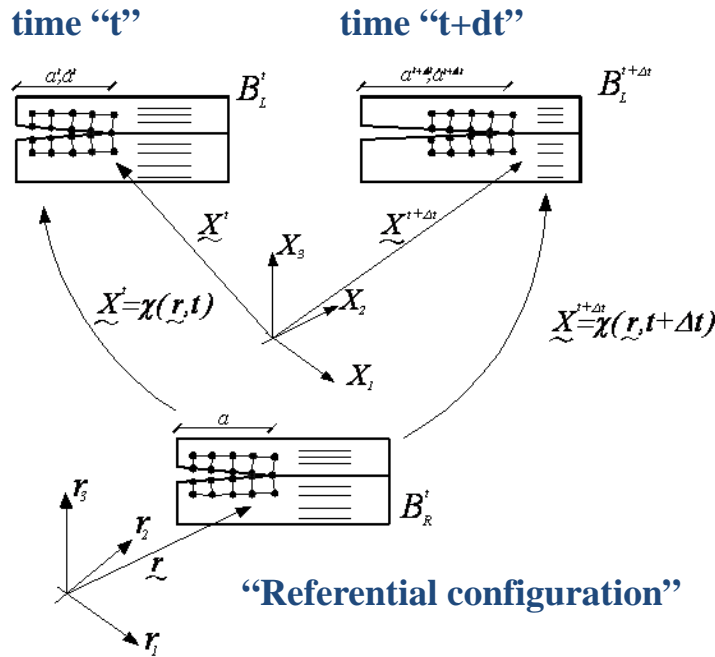
The Reference configuration is fixed and independent of any placement of the material body



BASICS OF MOVING MESH STRATEGY: ARBITRARY-LAGRANGIAN EULERIAN FORMULATION

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Physical quantities:

$$\underline{v} = \frac{d}{dt} \varphi(\underline{X}, t) \Big|_{\underline{X}}, \quad \underline{X}' = \frac{d}{dt} \chi(\underline{r}, t) \Big|_{\underline{r}}$$

“Material”

“Referential”



$$\underline{\&} = \underline{f}' - \underline{X}' \frac{d}{d\underline{X}} f(\underline{X}, t)$$

Time derivative rule

Physical fields in ALE formulation

$$\underline{\&} = \underline{u}'' - 2 \nabla_x \underline{u}' \cdot \underline{X}' - \nabla_x \underline{u} \underline{X}'' + \nabla_x (\nabla_x \underline{u}) \underline{X}' \underline{X}' + \nabla_x \underline{u} \nabla_x \underline{X}' \underline{X}'$$

“Material accel.”

$$\nabla_x \underline{u} = \nabla_r \underline{u} \underline{J}^{-1}$$

“Grad. transform.”

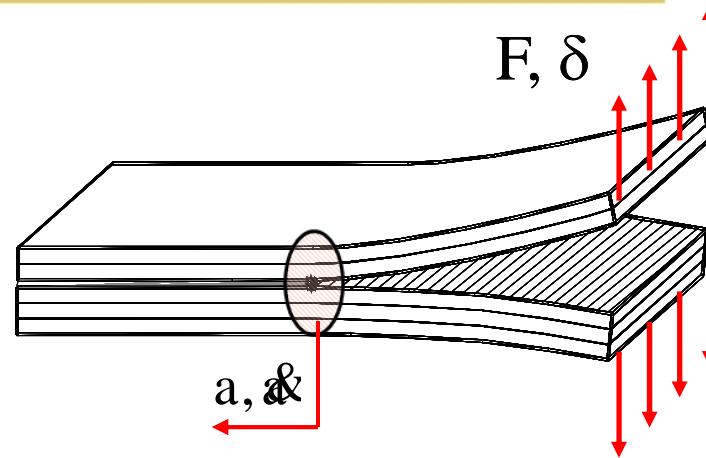
$$\det \underline{J} \neq 0$$

“one-to-one relationship”



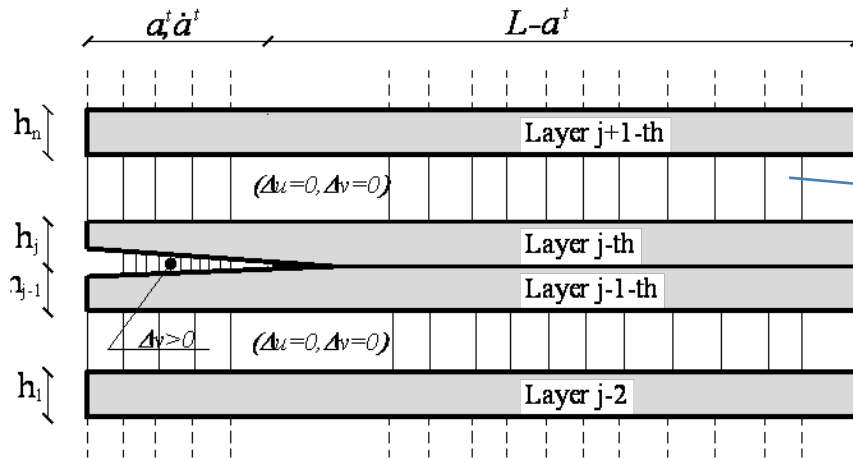
DESCRIPTION OF THE DELAMINATION MODEL

- Multi-layer Modeling**
- 2D Kinematic formulation**
- The laminate is divided into n mathematical layer representing the stacking sequence**



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- Compatibility equations LMM:**

$$\Delta u_i = u_{i+1} - u_i = 0, \quad \Delta v_i = v_{i+1} - v_i = 0,$$

“undelaminated interfaces”

$$\Delta v_i = v_{i+1} - v_i \geq 0,$$

“delaminated interfaces”



ERR RATE EVALUATION : J-INTEGRAL APPROACH

Revision of the J-integral Dec. procedure (Rigby & Aliabady, 1998, Greco & Lonetti, 2009))

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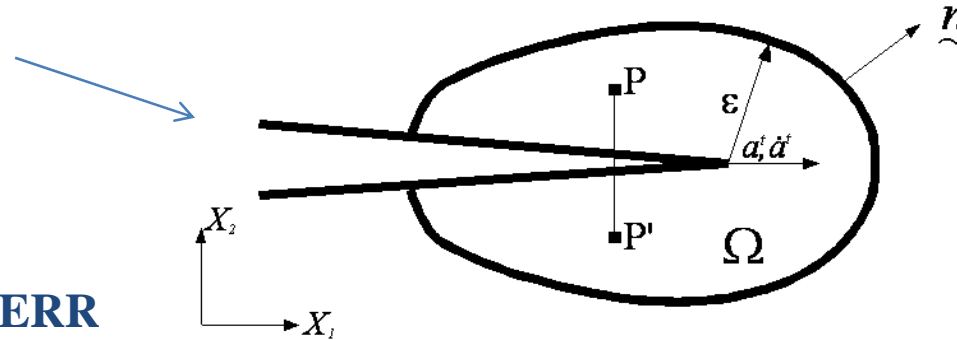
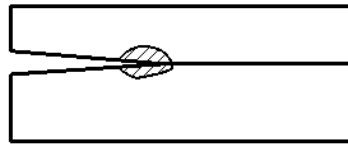
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Expressions of the ERR

$$J = \lim_{\varepsilon \rightarrow 0} \int_{\Omega} (W + K) n_1 - t \frac{\partial u}{\partial X_2} ds$$

$$J = \int_{\partial\Omega} (W + K) n_1 - t \frac{\partial u}{\partial X_2} ds + \int_{\Omega} \left[\rho \left(\frac{\partial u}{\partial x} - f \right) \nabla u - \rho \frac{\partial u}{\partial x} \right] dA$$

“Path independent”

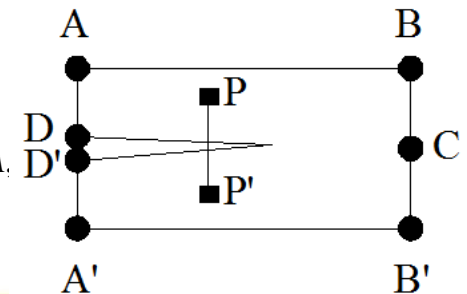
(Nishioka, T, 2001)

Decomposition of the ERR into symmetric and antisymmetric fields

$$J_I = G_I = \int_{\partial\Omega} \left[(W^S + K^S) n_1 - \sigma_{ij}^S n_j \frac{\partial u^S}{\partial x} \right] ds + \int_{\Omega} \left[\rho \left(\frac{\partial u^S}{\partial x} - f^S \right) \nabla u^S - \rho \frac{\partial u^S}{\partial x} \right] dA,$$

$$J_{II} = G_{II} = \int_{\partial\Omega} \left[(W^{AS} + K^{AS}) n_1 - \sigma_{ij}^{AS} n_j \frac{\partial u^{AS}}{\partial x} \right] ds + \int_{\Omega} \left[\rho \left(\frac{\partial u^{AS}}{\partial x} - f^{AS} \right) \nabla u^{AS} - \rho \frac{\partial u^{AS}}{\partial x} \right] dA,$$

(Lonetti et al., 2010)

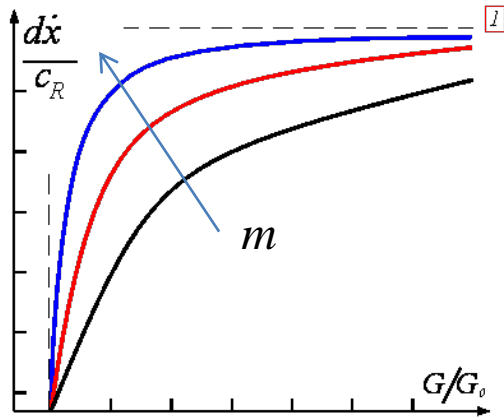


DYNAMIC CRACK PROPAGATION ANALYSIS: GROWTH CRITERION

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Crack growth criterion



$$G_D = \frac{G_0}{1 - \left(\frac{c_t}{V_R}\right)^m}$$

Material parameter
 “Critical value of the ERR”
 (Freund, 1990; Ravi-Chandar, 2004)

$$c \rightarrow V_R \quad G_D(c_t) = \infty$$

$$c \rightarrow 0 \quad G_D(c_t) = G_0(0)$$

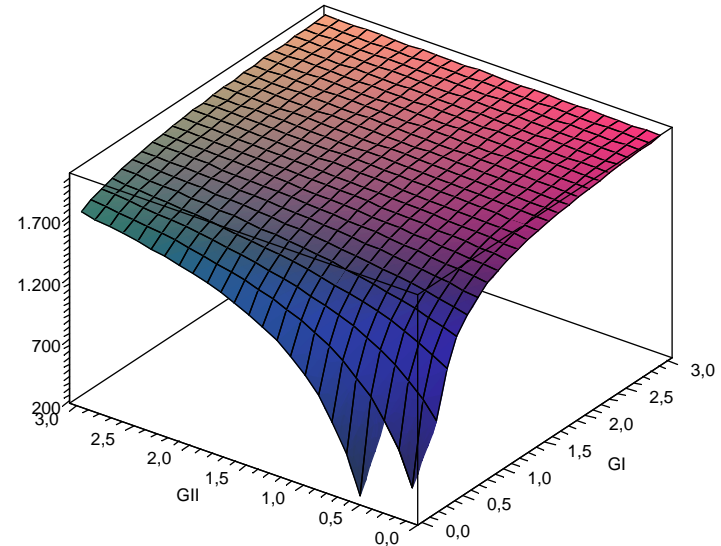
“Rayleigh wave speed”
 “initiation value”

1) Mixed mode crack growth criterion

$$g_f = \frac{G_I}{G_{ID}(c_t)} + \frac{G_{II}}{G_{IID}(c_t)} - 1 \leq 0$$

Material parameter

$$G_{ID}(c_t) = \frac{G_{0I}}{1 - \left(\frac{c_t}{V_R}\right)^m}, \quad G_{IID}(c_t) = \frac{G_{0II}}{1 - \left(\frac{c_t}{V_R}\right)^m}$$



DESCRIPTION OF THE DELAMINATION MODEL IN THE REFERENTIAL CONFIGURATION

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☐ Governing Equations: “Principle of d’Alembert”

$$\sum_{i=1}^n \int_{V_i} \sigma \delta \nabla u dV + \sum_{i=1}^n \int_{V_i} \rho \delta u dV = \sum_{i=1}^n \int_{\Omega_i} t \delta u dA + \sum_{i=1}^n \int_{V_i} f \delta u dV$$

Internal work

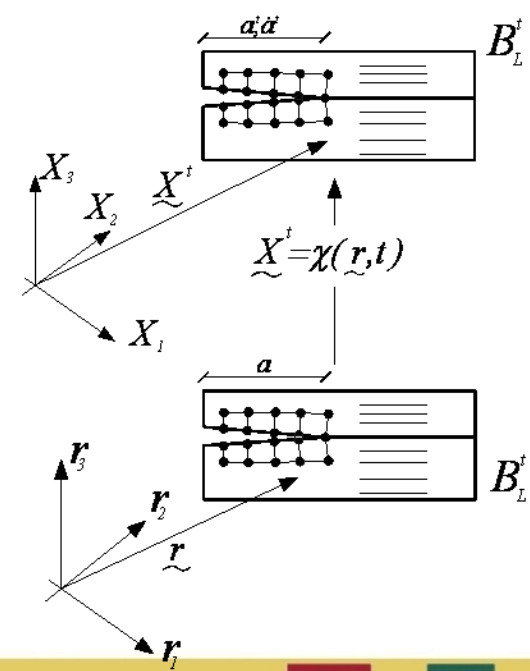
External work

➔
$$\sum_{i=1}^n \int_{V_i} \sigma \delta \nabla u dV = \sum_{i=1}^n \int_{V_{ri}} C(\nabla_r u J^{-1}) \delta(\nabla_r u J^{-1}) \det(J) dV_r$$

: Jacobian

➔
$$\sum_{i=1}^n \int_{V_i} \rho \delta u dV = \sum_{i=1}^n \int_{V_{ri}} \rho [u'' - 2 \nabla_r u' J^{-1} \cdot X' - (\nabla_r u J^{-1}) \cdot X'' + \nabla_r (\nabla_r u J^{-1}) J^{-1} X' X' + \nabla_r u J^{-1} \cdot (\nabla_r X J^{-1}) X'] \delta u \det(J) dV_r$$

➔
$$\sum_{i=1}^n \int_{\Omega_i} t \delta u dA + \sum_{i=1}^n \int_{V_i} f \delta u dV = \sum_{i=1}^n \int_{\Omega_{ri}} t \delta u \det(\bar{J}) d\Omega_r + \sum_{i=1}^n \int_{V_{ri}} f \delta u \det(J) dV_r$$



MOVING MESH METHOD: FUNDAMENTAL EQUATIONS

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ALE formulation to describe mesh motion

$$\rightarrow \nabla_{X_{0\%}}^2 \Delta X_1 = 0, \quad \nabla_{X_{0\%}}^2 \Delta X_2 = 0.$$

$$\Delta X_1 = X_1 - r_1 \quad \Delta X_2 = X_2 - r_2$$

“Mesh displacements of nodes
Should be regular”

Mesh regularization technique
“Winslow Smoothing method”

Minimize the mesh warping

Boundary conditions

$$\rightarrow (\Delta X_1^j = 0, \Delta X_2^j = 0) \quad \text{on } \Omega_1 \cup \Omega_2,$$

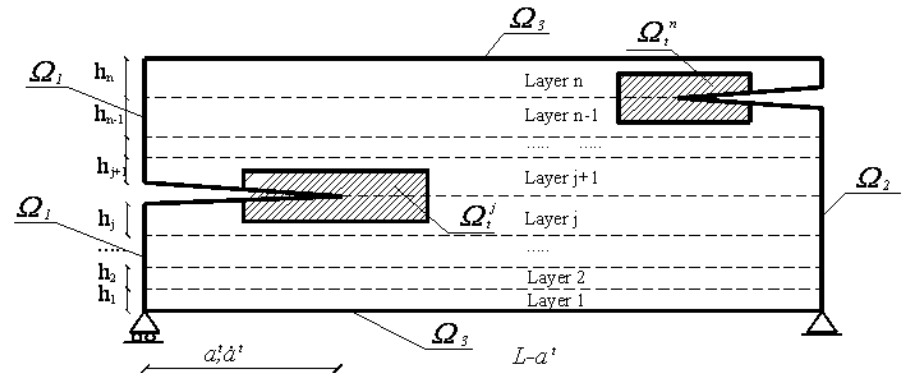
$$\Delta X_2^j = 0 \quad \text{on } \Omega_3 \cup \Omega_4$$

$$\Delta X_1^{j,j} = 0 \Leftrightarrow \text{if } g_f^j < 0 \text{ on } \Omega_t^j,$$

$$\Delta X_1^{j,j} = c_t \Leftrightarrow \text{if } g_f^j \geq 0 \text{ on } \Omega_t^j,$$

$$\Delta X_2^{j,j} = 0 \text{ on } \Omega_t^j$$

$$\Delta X_1^j(0) = 0, \Delta X_2^j(0) = 0, \Delta X_1^{'j}(0) = 0, \Delta X_2^{'j}(0) = 0$$



VARIATIONAL FORMULATION AND FE IMPLEMENTATION

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Weak forms: coupled equations for the ALE and PS formulations:

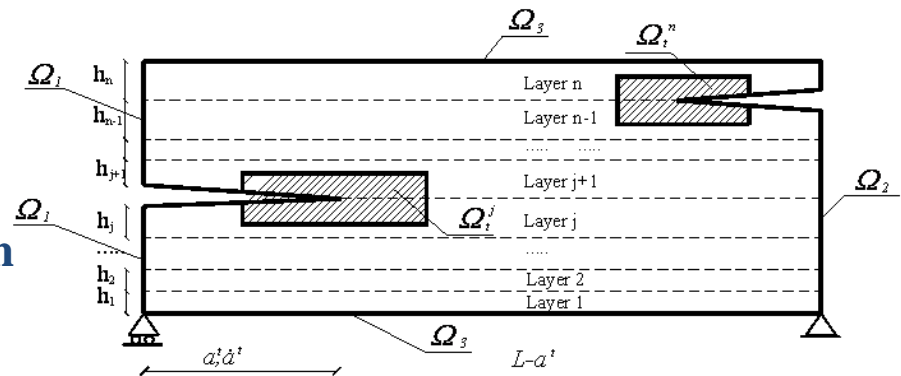
$$\begin{aligned} & \rightarrow \sum_{i=1}^n \int_{V_{ri}} C(\nabla_r u J^{-1}) \delta(\nabla_r u J^{-1}) \det(J) dV_r + \sum_{i=1}^n \int_{V_{ri}} \rho [u'' - 2 \nabla_r u' J^{-1} \cdot X' - (\nabla_r u J^{-1}) \cdot X'' + \\ & + \nabla_r (\nabla_r u J^{-1}) J^{-1} X' X' + \nabla_r u J^{-1} \cdot (\nabla_r X J^{-1}) X'] \delta u \det(J) dV_r \quad \text{PS} \\ & = \sum_{i=1}^n \int_{\Omega_{ri}} t \delta u \det(\bar{J}) d\Omega_r + \sum_{i=1}^n \int_{V_{ri}} f \delta u \det(J) dV_r \end{aligned}$$

$$\rightarrow \int_{V_r} (\nabla_r \Delta X J^{-1}) \cdot (\nabla_r w J^{-1}) \det(J) dV_r + \int_{\Omega_r} [\delta \lambda (X' - c) i + \lambda \delta X i] (\bar{J}) ds = 0, \quad \text{ALE}$$

Explicit equations for PS+ALE

Implicit equations for the crack growth

Crack growth criterion



VARIATIONAL FORMULATION AND FE IMPLEMENTATION

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FE approximation by “Comsol Multiphysics”:

- ➔ Quadratic Lagrangian interpolation functions for displacements, velocity and acceleration fields
- ➔ Quadratic Lagrangian interpolation functions for mesh points displacements

FE equations

$$\sum_{i=1}^n M_{\%}^i U_j'' + \sum_{i=1}^n C_{\%}^i U_j' + \sum_{i=1}^n (K_{\%}^j + K_{\%}^{0i} + K_{\%}^{1i} + K_{\%}^{2i}) U_j + \sum_{i=1}^n T_{\%}^j + \sum_{i=1}^n P_{\%}^j = 0$$

$$W_{\%} \cdot \Delta X_{\%} + Q_{\%} \cdot \Delta X'_{\%} + L_{\%} = 0,$$

Solution Procedure

- ➔ Implicit time integration scheme based on variable-step-size backward differentiation formula

Non Linear Equations System



Iterative-incremental Solving procedure



RESULTS: VALIDATION OF THE STRUCTURAL MODEL

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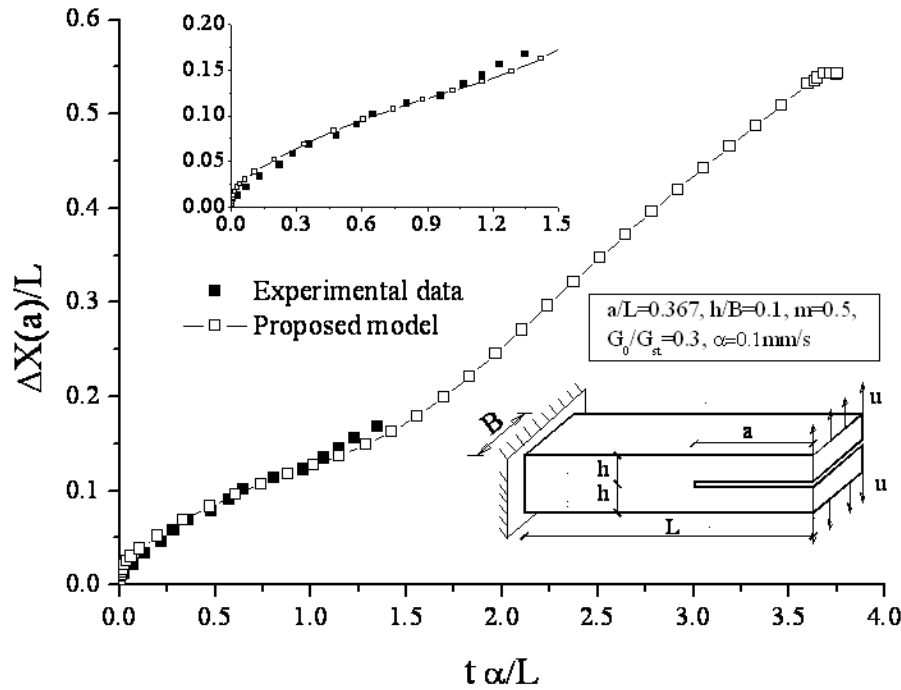
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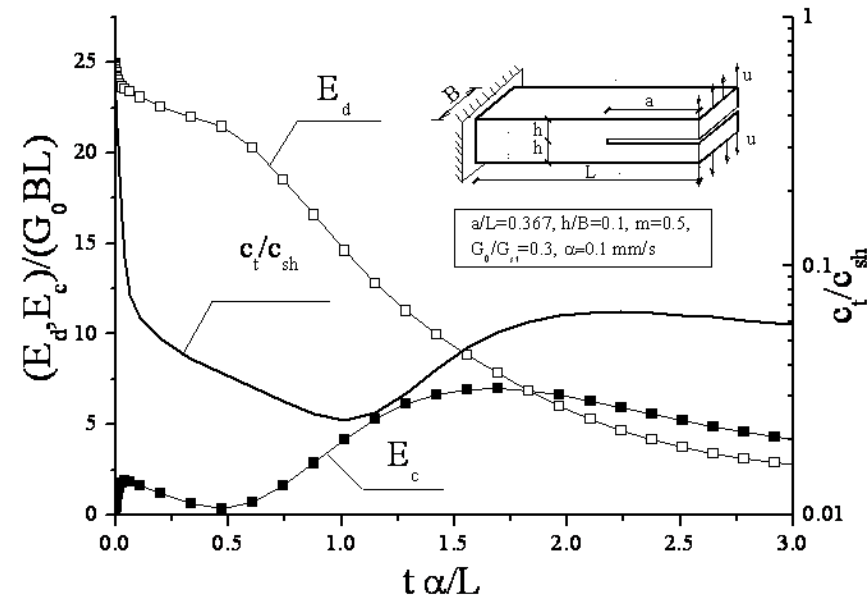
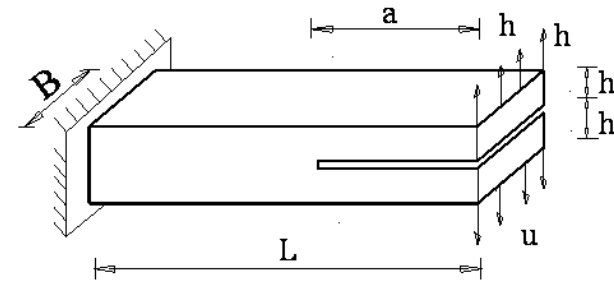


DCB mode I loading scheme

Comparisons with experimental data

AS 3501-6 Graphite/Epoxy

(Guo C, Sun CT., 1998)

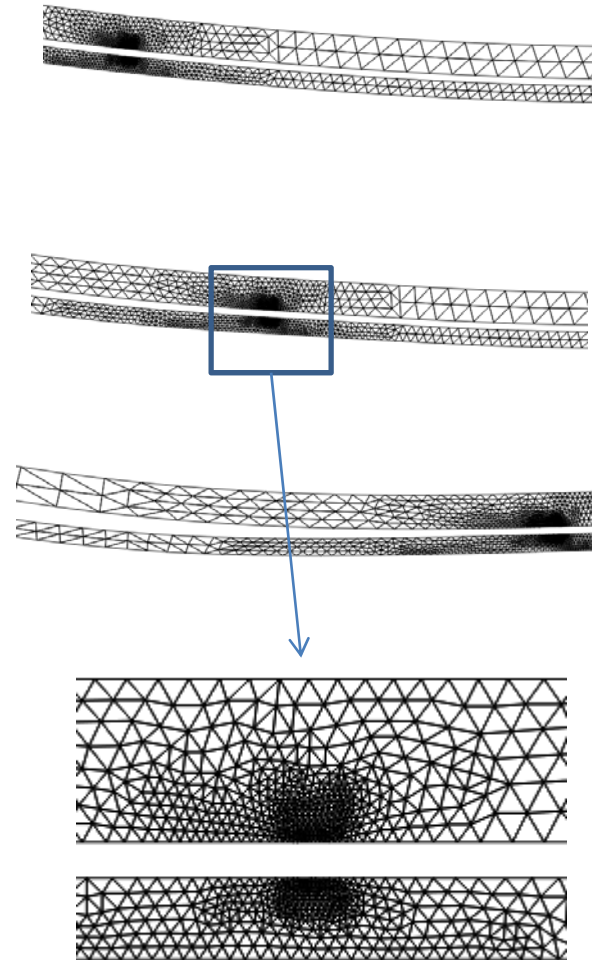
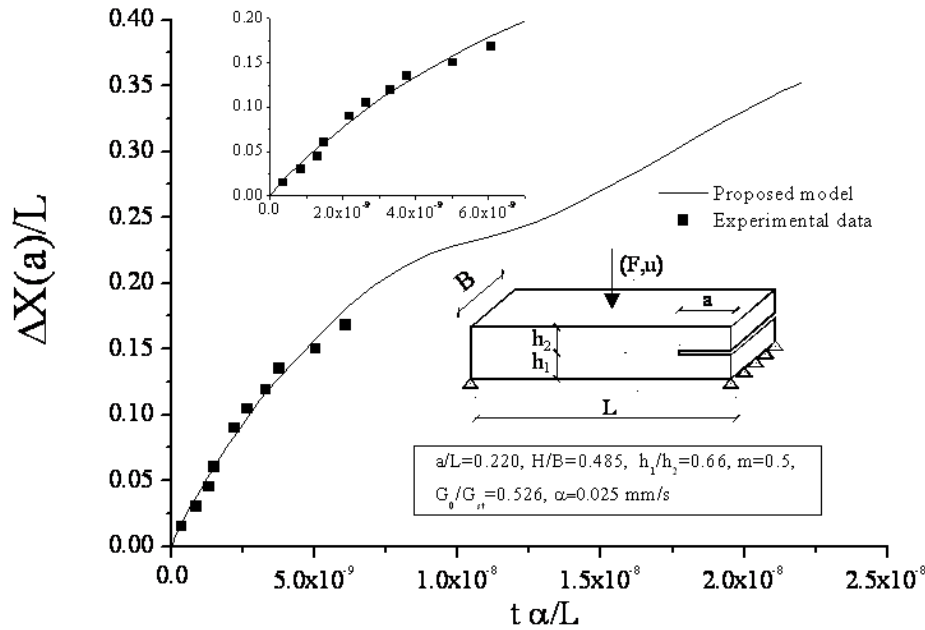


RESULTS: VALIDATION OF THE STRUCTURAL MODEL

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Time incrementation



- ▣ DCB mode I loading scheme
- ▣ Comparisons with experimental data
- ▣ AS 3501-6 Graphite/Epoxy
([Tsai JL, Guo C, Sun CT. , 2001](#))



RESULTS : MIXED MODE ANALYSIS

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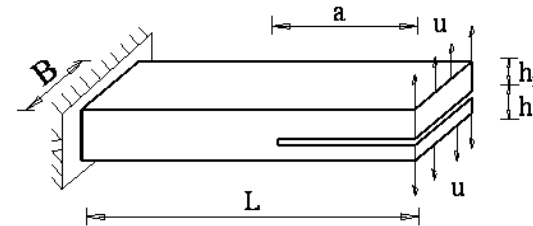
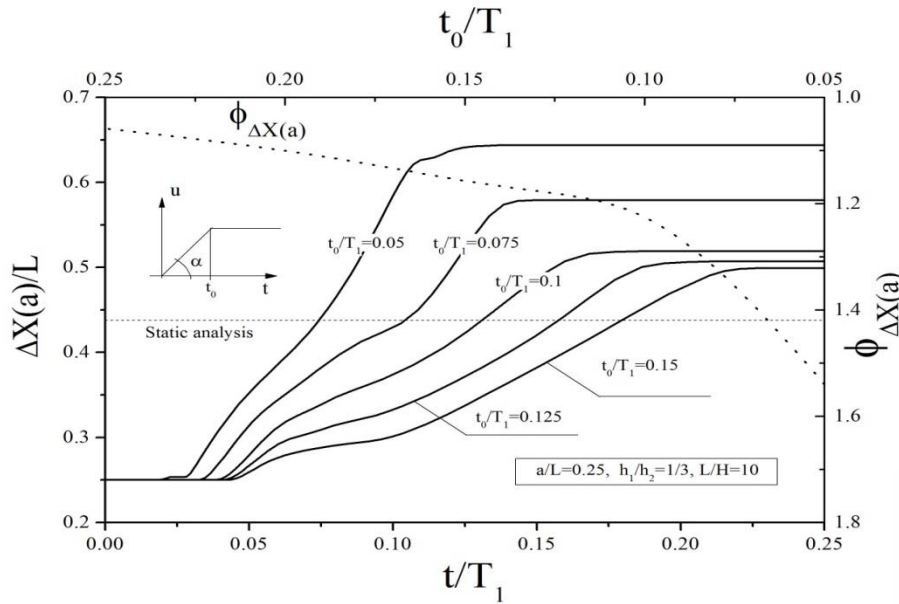
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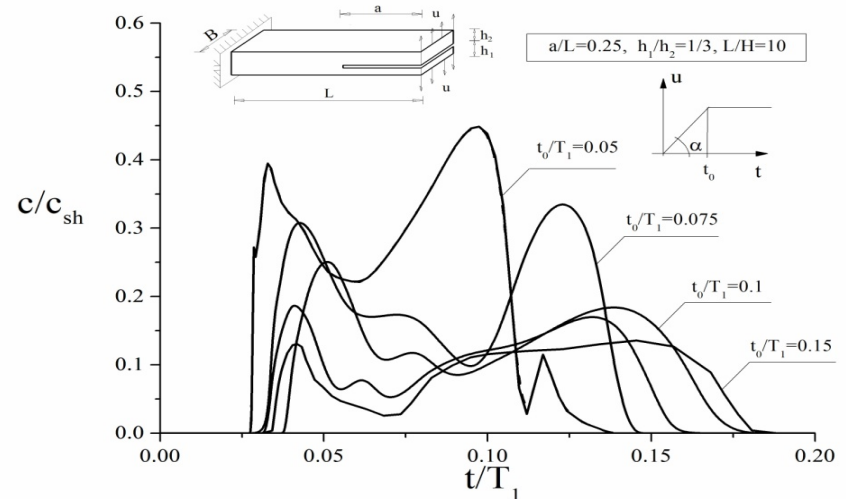
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-  Influence of the Loading Rate
-  DAFs of the crack tip length
-  Evolution of the crack tip speed



RESULTS : MULTIPLE DELAMINATIONS

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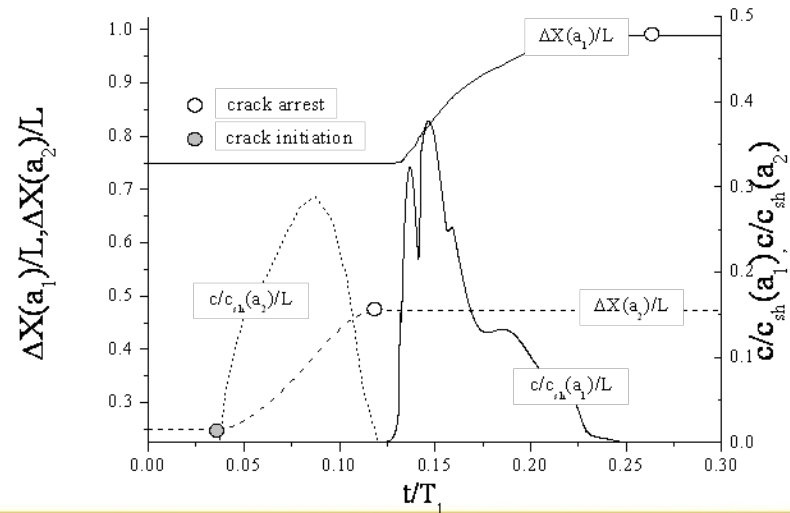
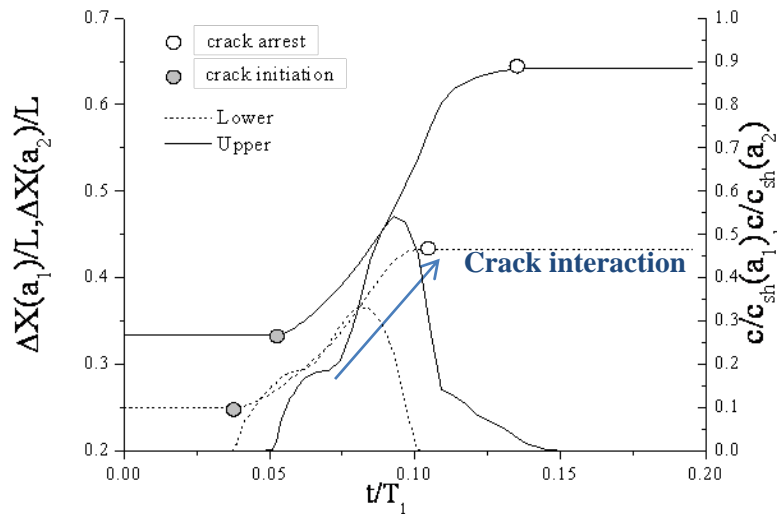
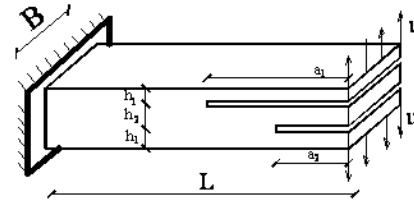
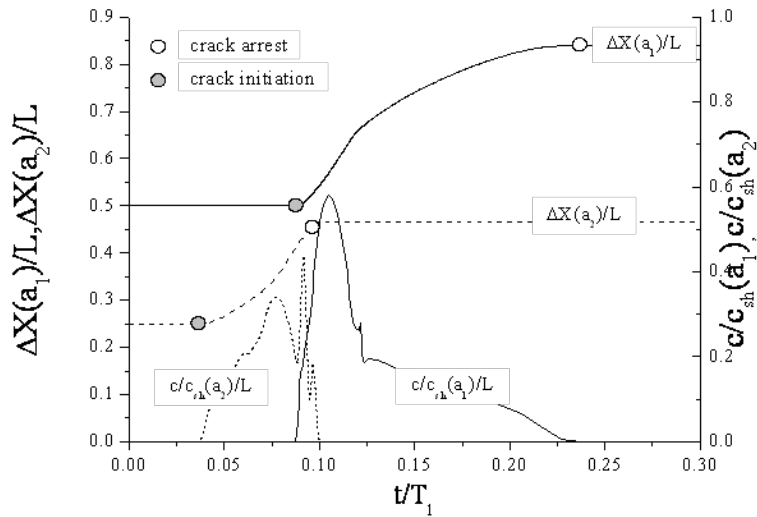
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CONCLUDING REMARKS

- ❏ A delamination model for general loading conditions based on moving mesh methodology and fracture mechanics is proposed.
- ❏ New expressions of the ERR mode components based on the J-integral decomposition procedure are derived.
- ❏ Comparisons with experimental data are proposed to validate the delamination modelling
- ❏ The analyzed parametric study shows that delamination phenomena are quite influenced by the loading rate, inertial effects leading to high amplifications in the ERR prediction and the crack growth.

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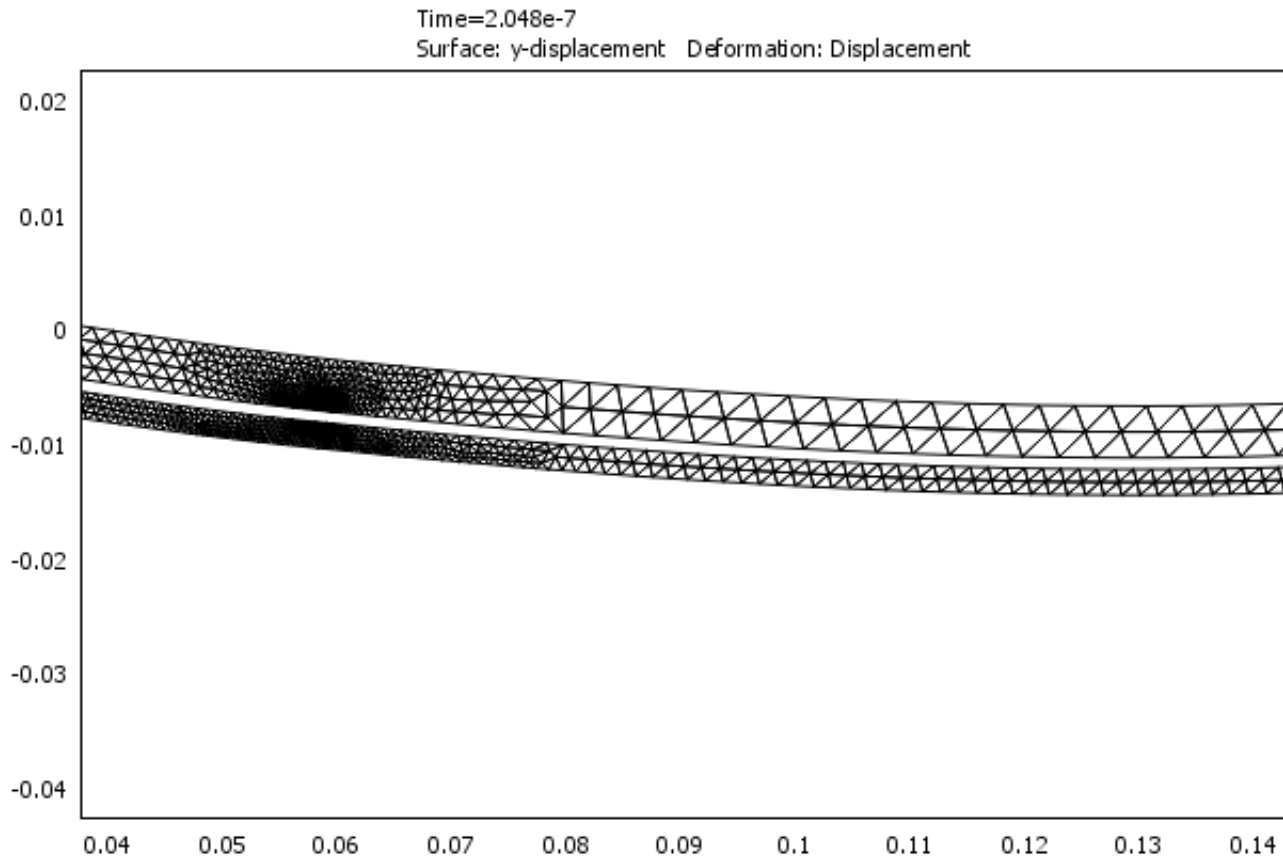
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GOVERNING EQUATIONS OF THE ALE-FM MODEL

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▣ Variational form of the structural model:

$$\sum_{i=1}^n \int_{V_{ri}} C(\nabla_r u J^{-1}) \delta(\nabla_r u J^{-1}) \det(J) dV_r + \sum_{i=1}^n \int_{V_{ri}} \rho [u'' - 2 \nabla_r u' J^{-1} \cdot X' - (\nabla_r u J^{-1}) \cdot X'' + \nabla_r (\nabla_r u J^{-1}) J^{-1} X' X' + \nabla_r u J^{-1} \cdot (\nabla_r X' J^{-1}) X'] \delta u \det(J) dV_r = \sum_{i=1}^n \int_{\Omega_{ri}} t \delta u \det(\bar{J}) d\Omega_r + \sum_{i=1}^n \int_{V_{ri}} f \delta u \det(J) dV_r$$

Unknown quantities: $(u, \Delta X, X, J)$ \longrightarrow **depend on the mesh configuration**

▣ Variational formulation of the ALE equation

$$\int_V \nabla_{\%}^X \Delta X_1 \nabla_{\%}^X w_1 dA + \int_{\Omega} [\delta \lambda (X'_1 - c_t) + \lambda \delta X'_1] ds = 0, \longrightarrow$$

$$\int_V \nabla_{\%}^X \Delta X_2 \nabla_{\%}^X w_2 dA = 0,$$

Crack growth criterion

LMM is utilized to impose the mesh speed at the delamination plane



RESULTS : MULTIPLE DELAMINATIONS

Influence of the Loading Rate

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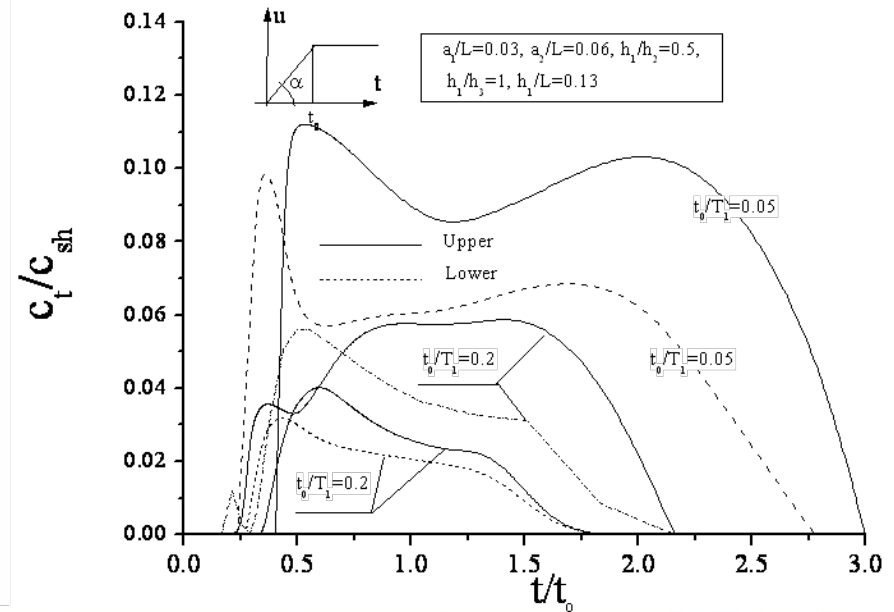
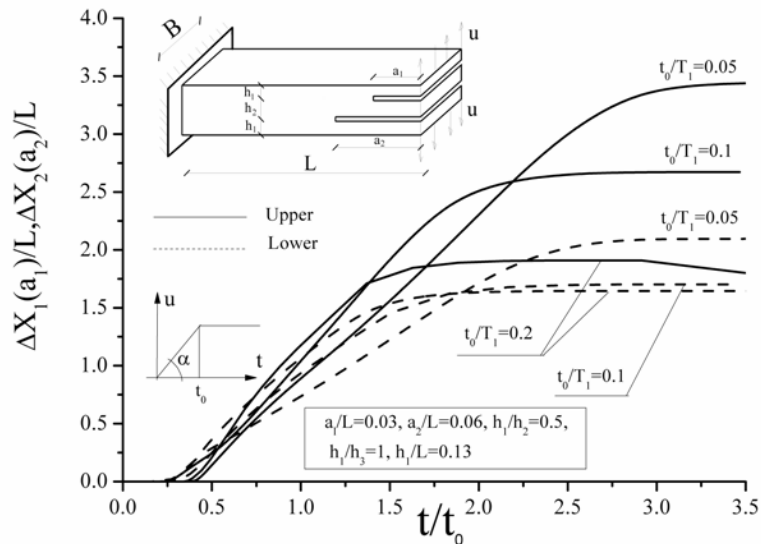
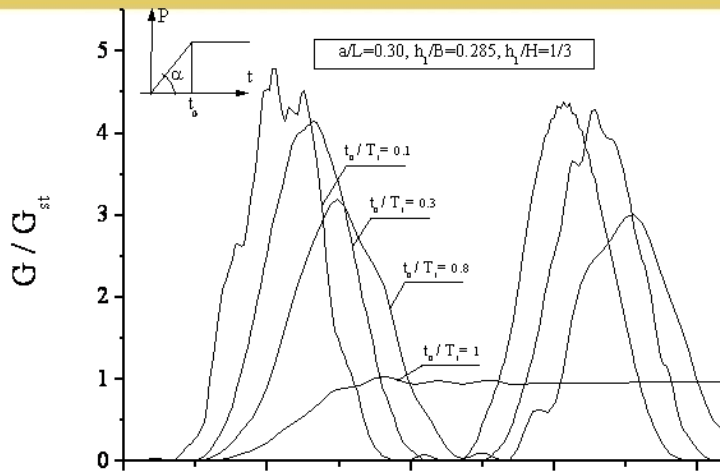
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Physical fields in ALE formulation

$$\frac{d}{dt} \mathbf{u} = \frac{d}{dt} \mathbf{u}' - \mathbf{X}' \frac{d}{dX} \mathbf{u}(\mathbf{X}, t) \longrightarrow \text{“Material velocity”}$$

“Material Configuration”

$$\frac{d}{dt} \mathbf{a} = \frac{d}{dt} \mathbf{u}'' - 2 \nabla_x \mathbf{u}' \cdot \mathbf{X}' - \nabla_x \mathbf{u} \mathbf{X}'' + \nabla_x (\nabla_x \mathbf{u}) \mathbf{X}' \mathbf{X}' + \nabla_x \mathbf{u} \nabla_x \mathbf{X}' \mathbf{X}'$$

“Material accel.”

Fundamental relationships between Material and Referential configurations

$$\nabla_x \mathbf{u} = \nabla_r \mathbf{u} \mathbf{J}^{-1} \longrightarrow \mathbf{J} = \begin{bmatrix} \frac{dX_1}{dr_1} & \frac{dX_1}{dr_2} \\ \frac{dX_2}{dr_1} & \frac{dX_2}{dr_2} \end{bmatrix}$$

“Referential Configuration”

$$\det \mathbf{J} \neq 0 \quad \text{“one-to-one relationship”}$$



RESULTS: EFFECT OF THE LOADING RATE

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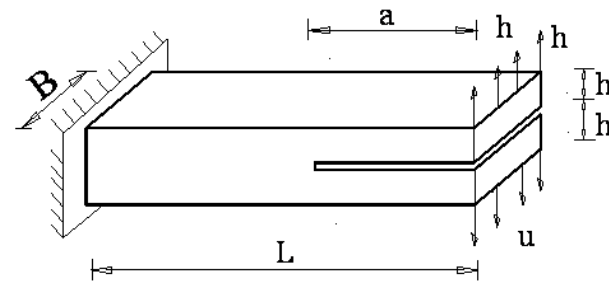
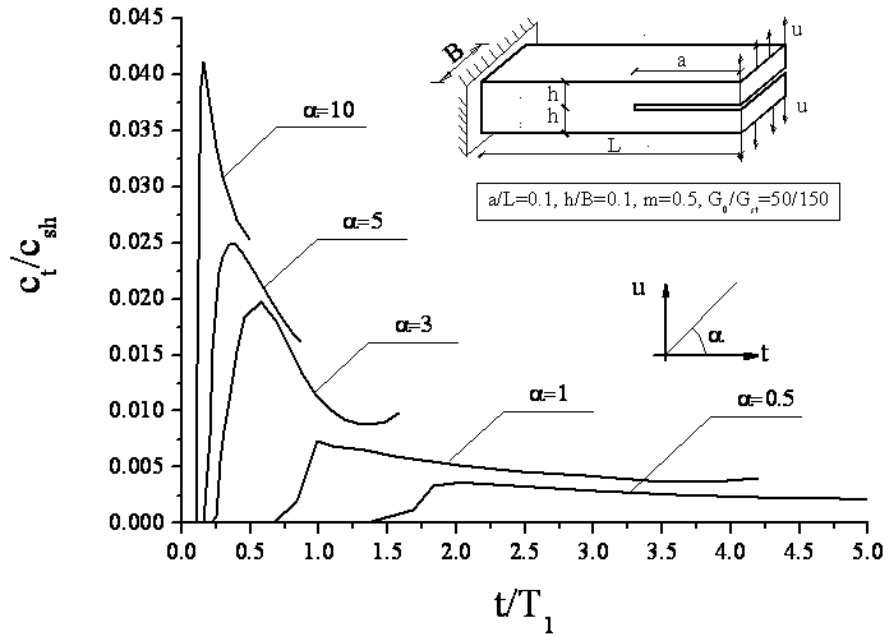
ALE MODEL




FORMULATION

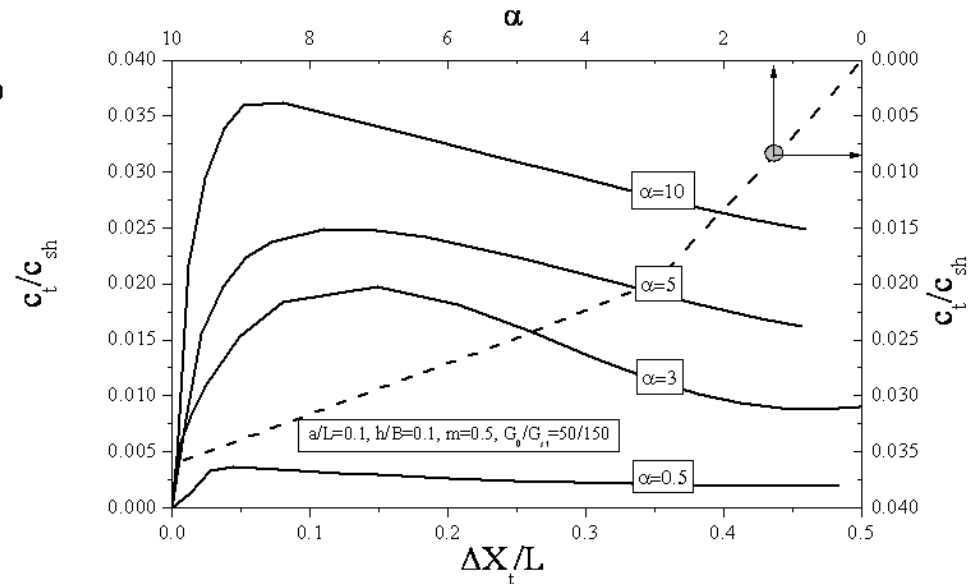
FE MODEL

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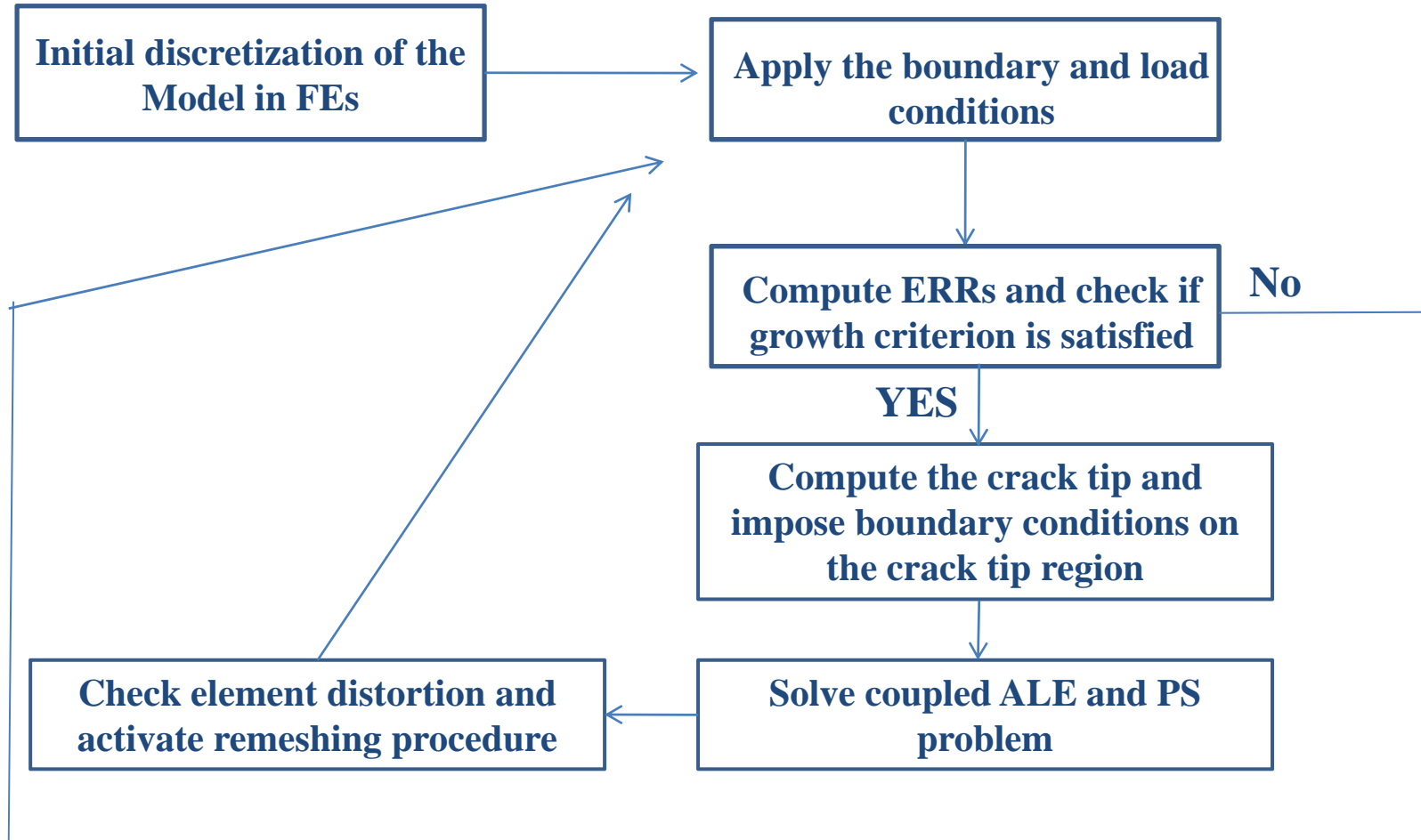
CONCLUSIONS



-  DCB mode I loading scheme
-  Influence of the loading rate
-  Evolution of the crack tip speed



SYNOPTICAL REPRESENTATION OF THE FEM ALGORITHM



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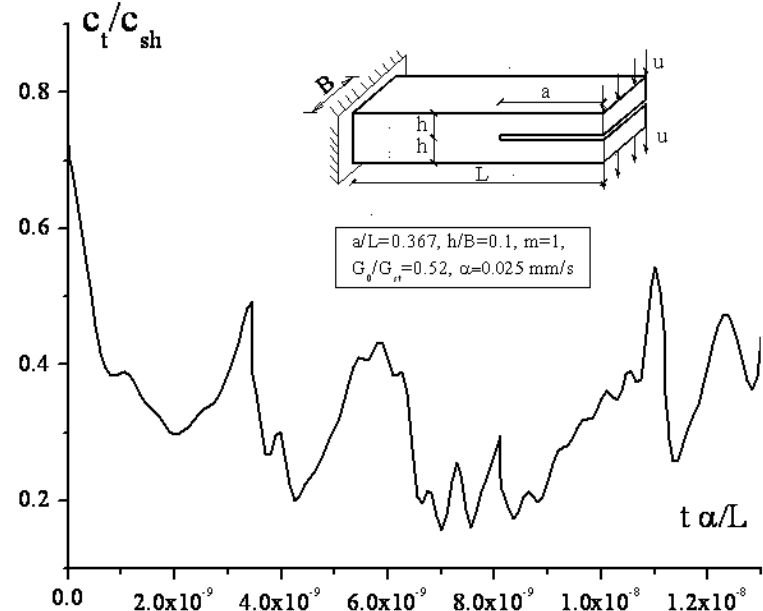
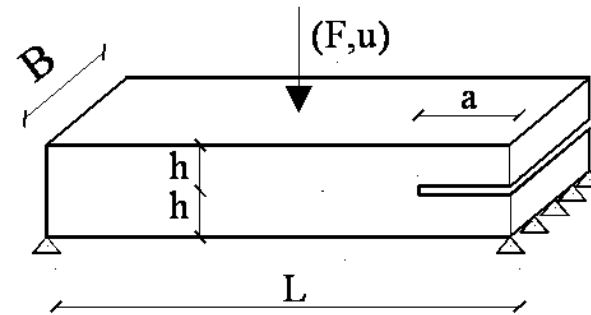
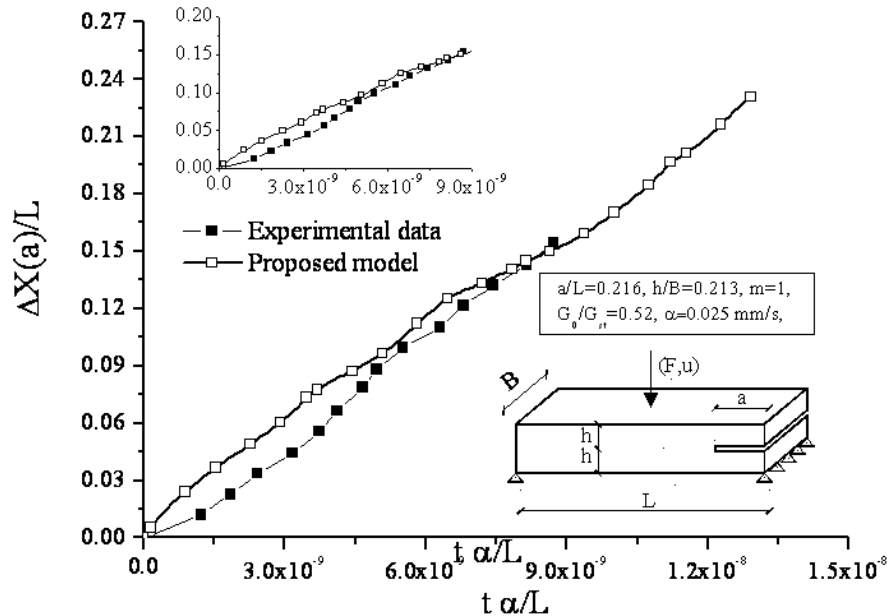
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RESULTS : MODE II ENF SCHEME

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-  ENF mode II loading scheme
-  Comparisons with experimental data
-  S2/8553 Glass/Epoxy



RESULTS : MODE II ENF SCHEME

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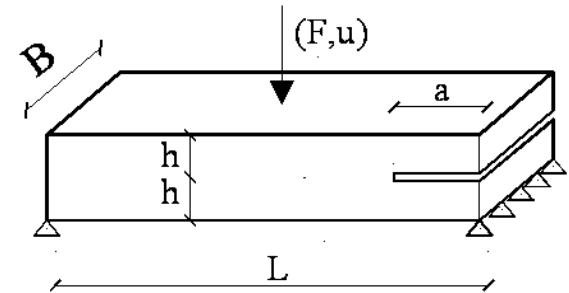
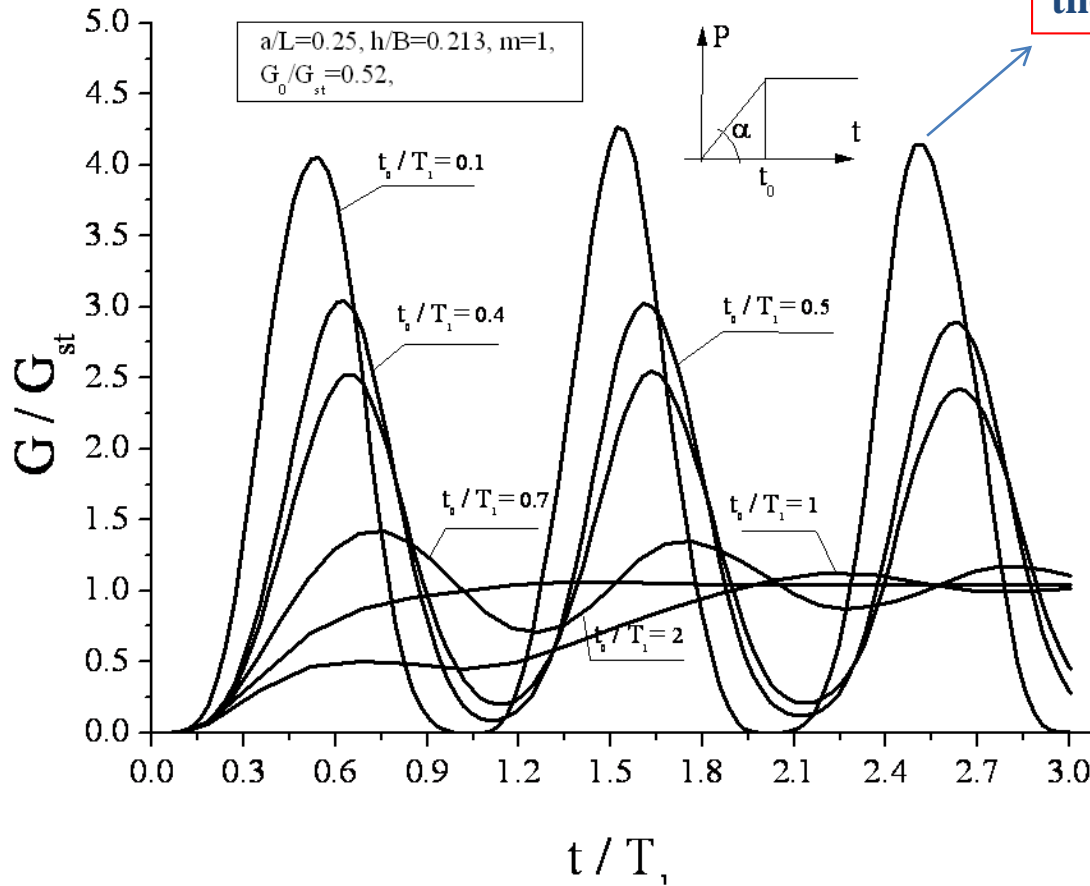
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RESULTS: VALIDATION OF THE STRUCTURAL MODEL

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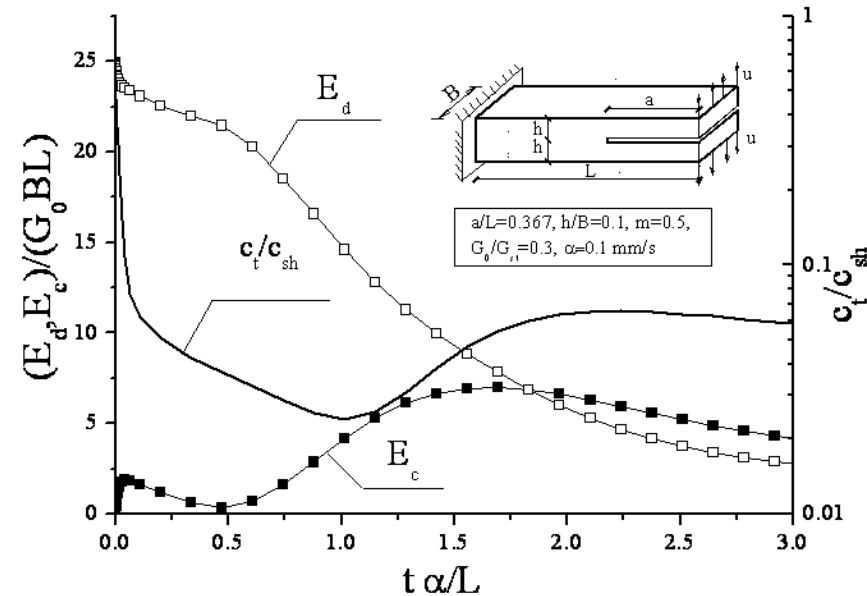
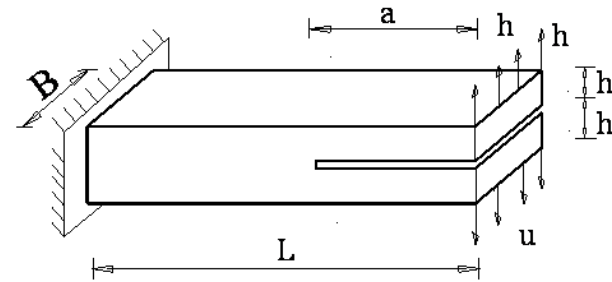
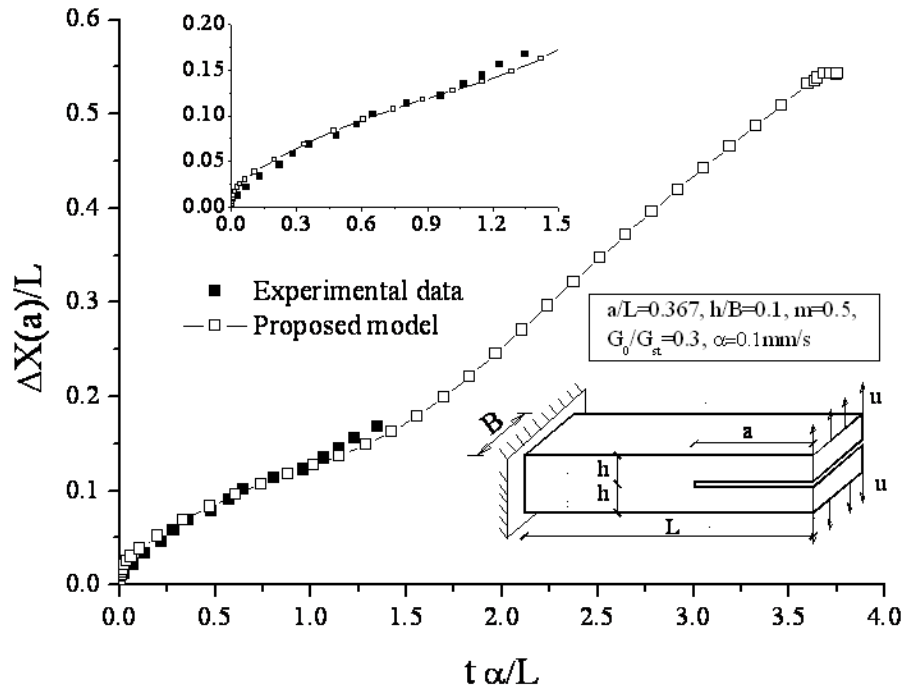
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- DCB mode I loading scheme
- Comparisons with experimental data
- AS 3501-6 Graphite/Epoxy



RESULTS : MIXED MODE ANALYSIS

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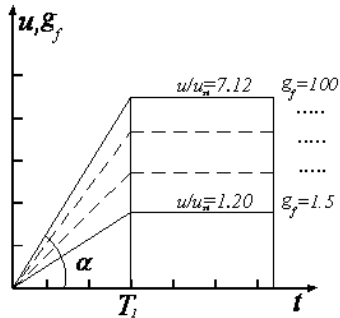
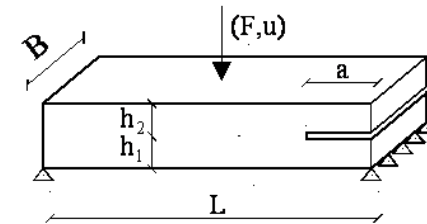
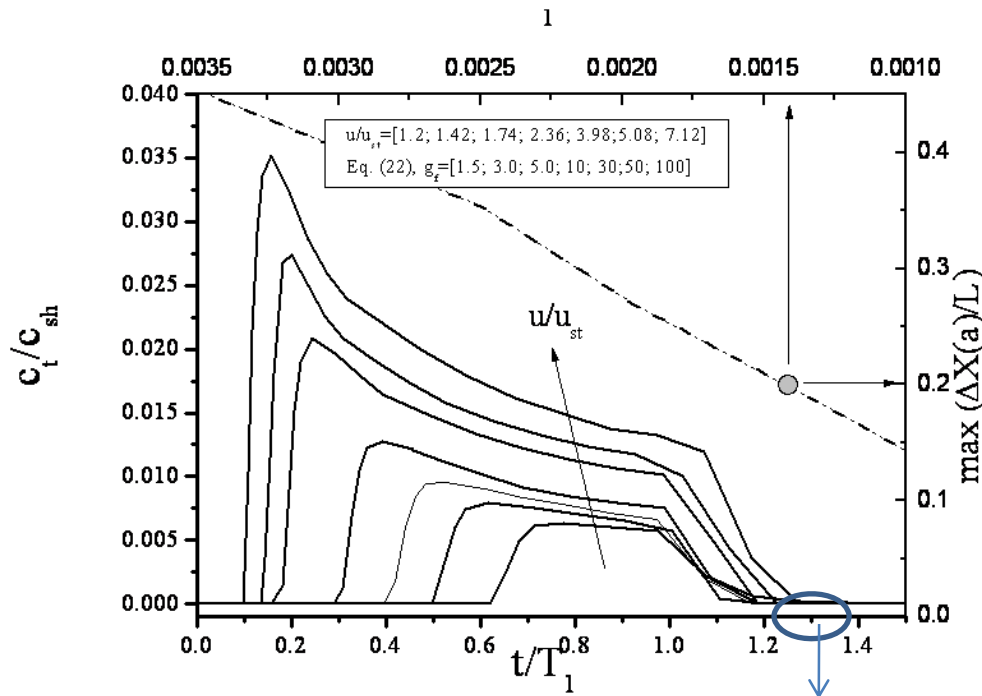
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Crack arrest phenomenon

Loading curves

